

FMEA
EMU FAILURE MODE, EFFECT ANALYSIS

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Date: 03/09/90

01/02/90 SUPERSEDES / /

EMERGENT

NAME P/N QTY	FUNCTION	FAILURE MODE & CAUSES		MISSION PHASE	FAILURE EFFECT	FAILURE DETECTION FLIGHT/GROUND	TIME TO EFFECT/ACTIONS		CRIT	REMARKS/ HAZARD	REF
		MISSION	PHASE				TIME	ACTION			
ELECTRICAL SIGNALS HANDLER, ITEM 152 SV780152-3 (1)	Electrically connects DCM to the CWS to provide status input signals, digital display signals, +5V, +10V, +/-4.2V power, SITE signals, motor tact signal. Electrically connects the DCM to the EVC to provide connections for hardline signals, mic signals, mic power, earphone signals and EOS signals.	152PM20s: Electrical open or short, one or more EVA display data (ASCII) lines.	PREEVA	END ITEM; Electrical open or short in one or more of the ASCII data lines from the CWS.	FLIGHT: Rel. Would be detected during periodic status checks or other failure display activation.	None.	3/10	A-PASS B-PASS C-PASS	Redundant path to None, the ventilation, CO2 removal, or pressurization components. In event of a short, each ASCII line is protected by a 330 ohm resistor in the CWS.		
	CAUSES:	CABLE CHAFING against connector shell or shield. Improper connector strain relief. Faulty connection between connector and lead wires.			CRE INTERFACE: All characters may not be properly displayed. Display may be illegible. Crew may not be able to read display properly.	GND/NO: Tee, PERU R-001, Para. 7.3.3.2.1.1.	TIME REQUIRED: N/A				
					MISSION: None.						
					CREW/VEHICLE: None for single failure. Possible loss of crewman with loss of CO2, oxygen or low vent flow.						

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ATTACHMENT
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